



European Thermodynamics Limited

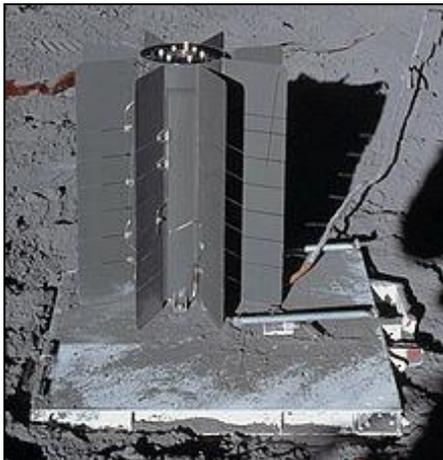
Intelligent Thermal Management

Thermoelectric Modules for Space Exploration Applications

Mark Robbins, Richard Tuley and Kevin Simpson.

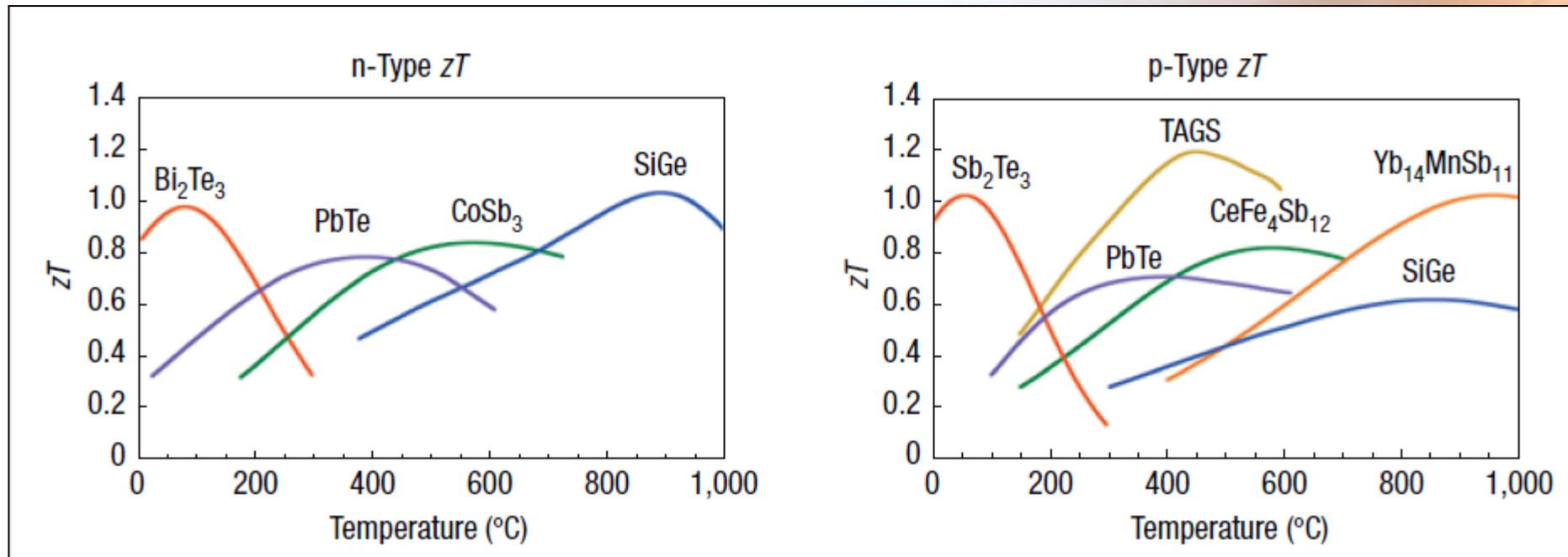
Background

- Designed for long missions where solar tech cannot power the craft reliably.
- SNAP (NASA) models were first RTGs to be launched using Plutonium-238 and Uranium-235 heat sources.
 - Silicon-germanium, lead telluride and tellurides of antimony, germanium and silver (TAGS).
 - High energy density heat sources.
 - Cassini (Saturn), Voyager (Outer Planets), Curiosity...



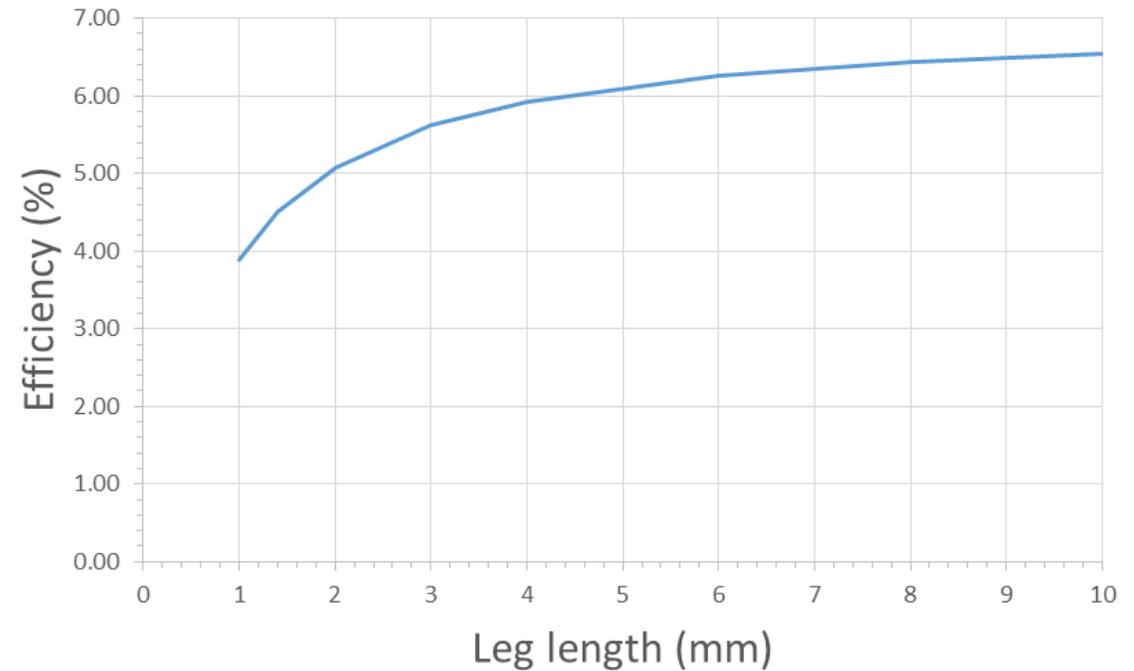
Background

- ESA system to use Americium241.
 - 80W thermal power.
 - With 5% system efficiency => 4W electrical.



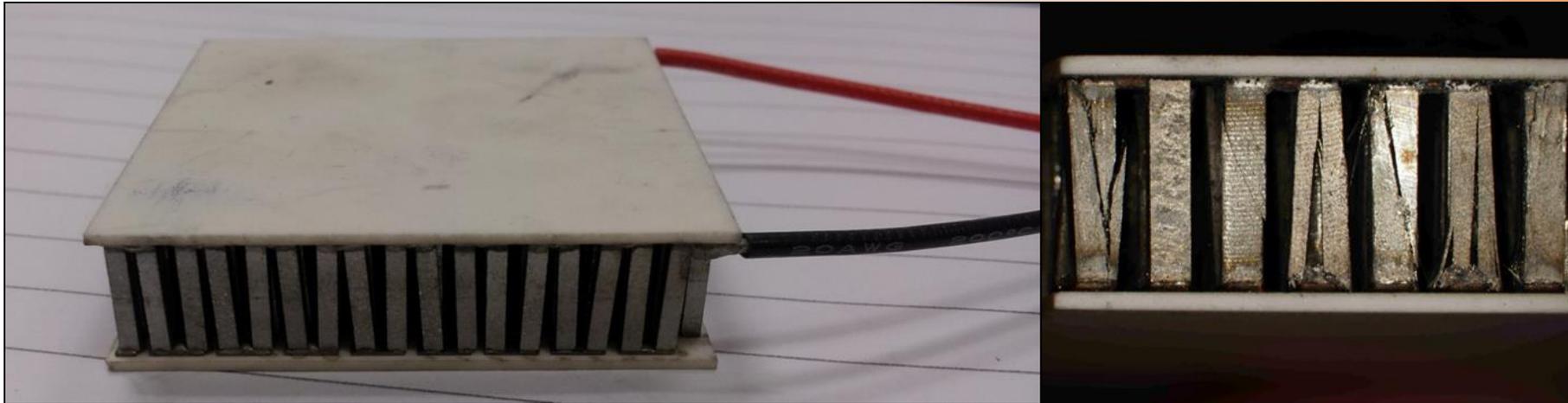
Module Requirements

- High efficiency.
- Many couples.
- Robust.
- Requires lower hot side temperature energy harvesting.
- Very low module thermal conductivity to maintain hot side temperature.



Challenges in Manufacture

- Taller legs and more couples = more difficult to build.
- Bismuth telluride is brittle.
 - Further complexity in build.
 - Structural strength could be an issue.



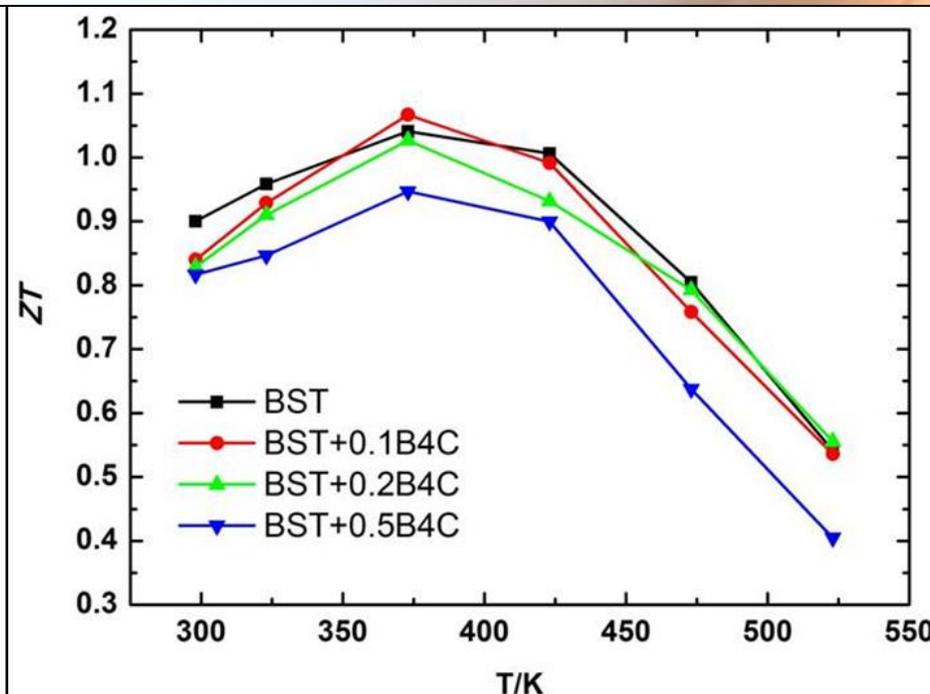
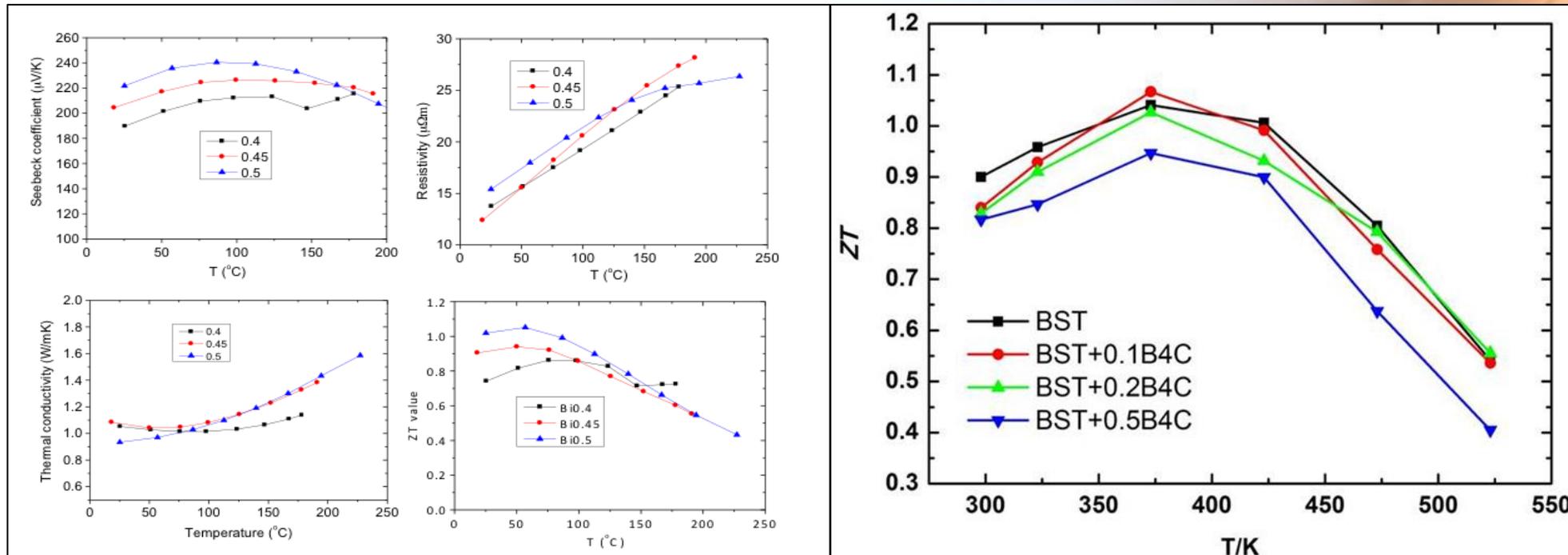
Materials and Manufacture

- To improve mechanical properties of bismuth telluride, three routes are compared:
 - Commercial zone melted.
 - SPS from nanopowders.
 - SPS from nanopowder with addition of boron carbide (B₄C).



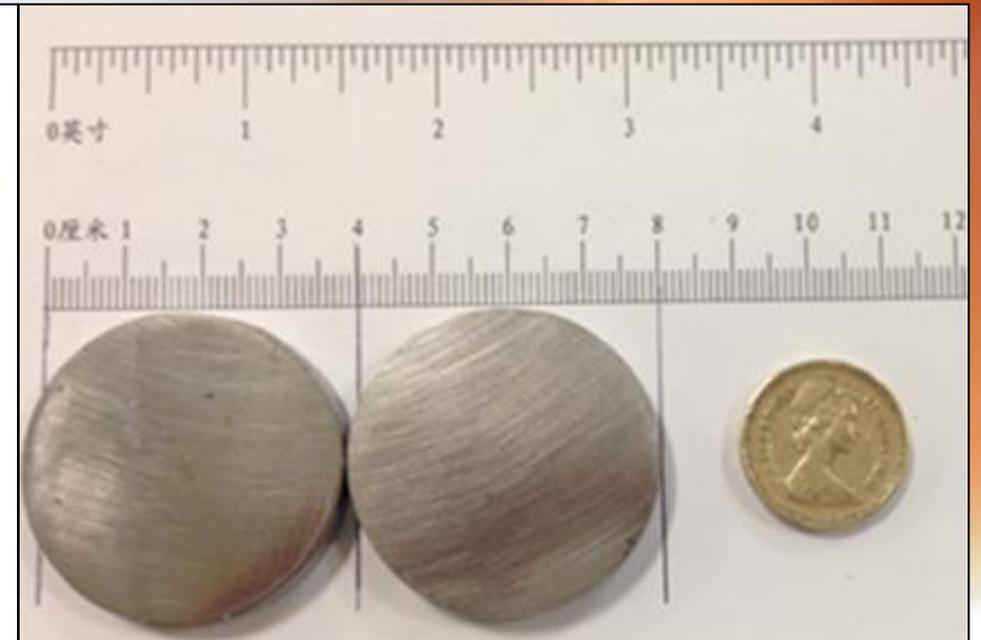
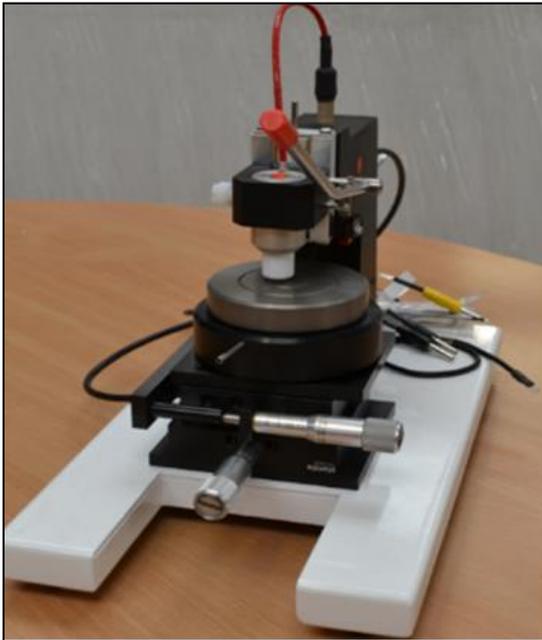
Materials and Manufacture

- P-type made by SPS manufacturing route.
- Addition of B4C to increase mechanical properties.



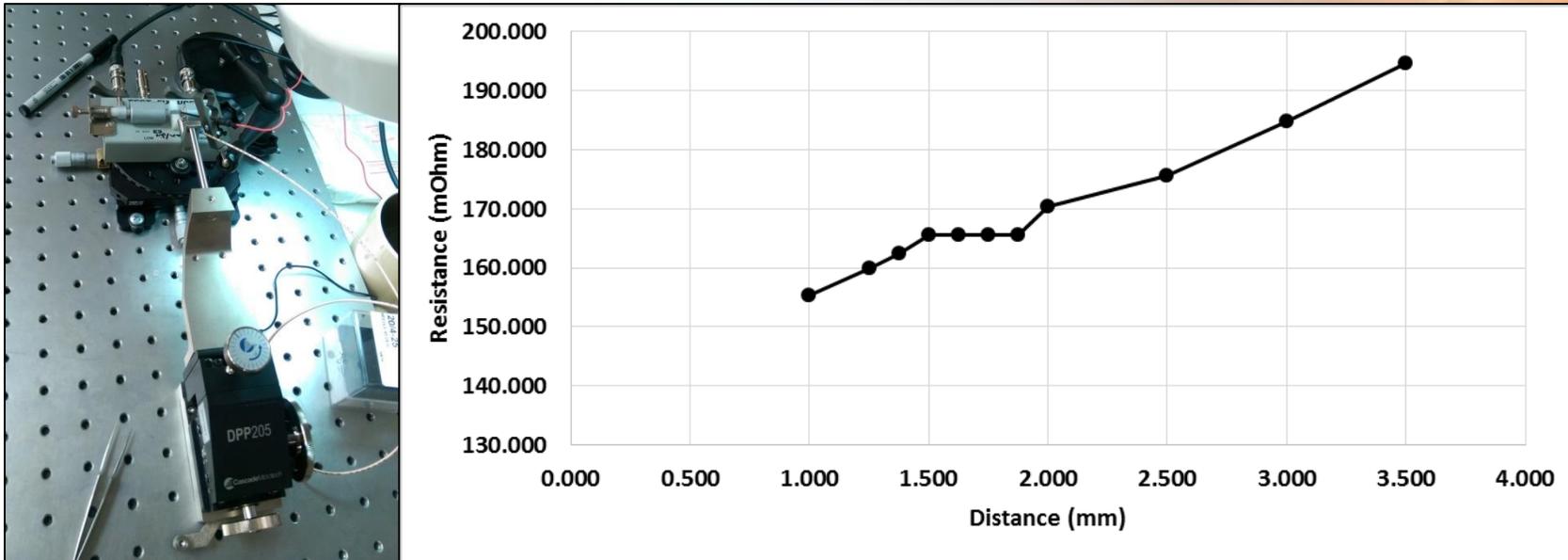
Puck Electrical Resistivity Mapping

- Room temperature measurements.
 - Quality and uniformity.

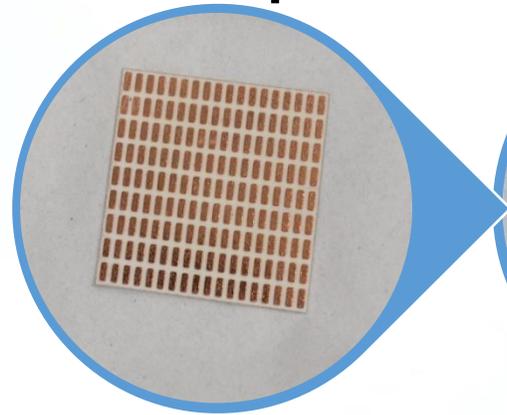


Joining Tests

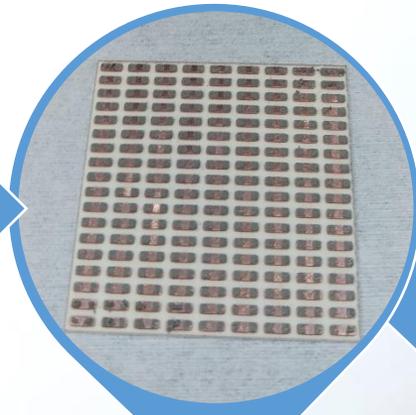
- Contact resistance $<10^{-5} \Omega \cdot \text{cm}^2$.
- Represents $<5\%$ of total resistance.



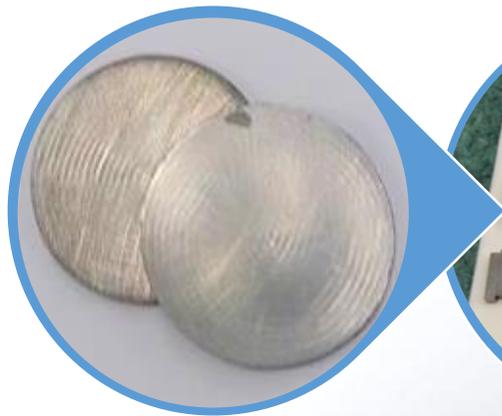
Production process



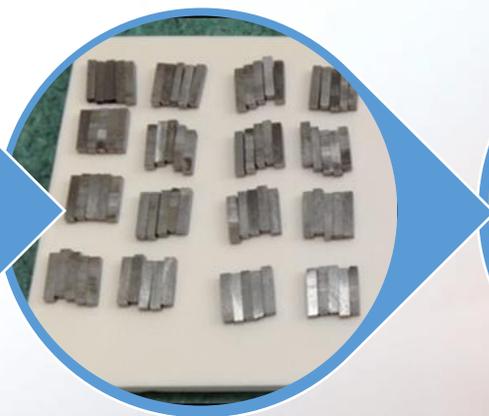
DBC ceramic



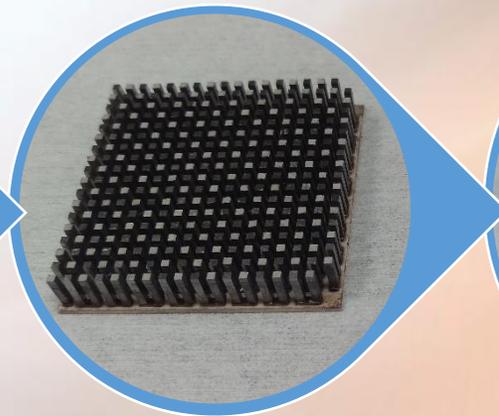
Solder deposition



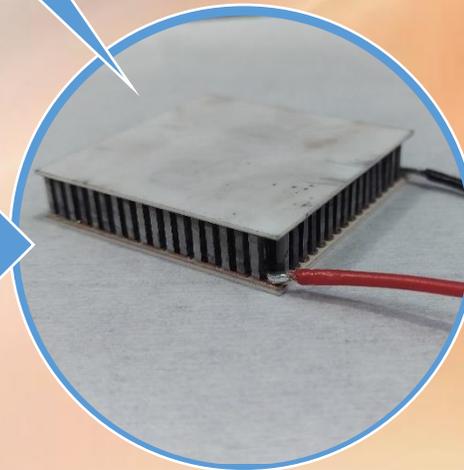
Material formulation
and consolidation



Grind, Metallisation
and cutting

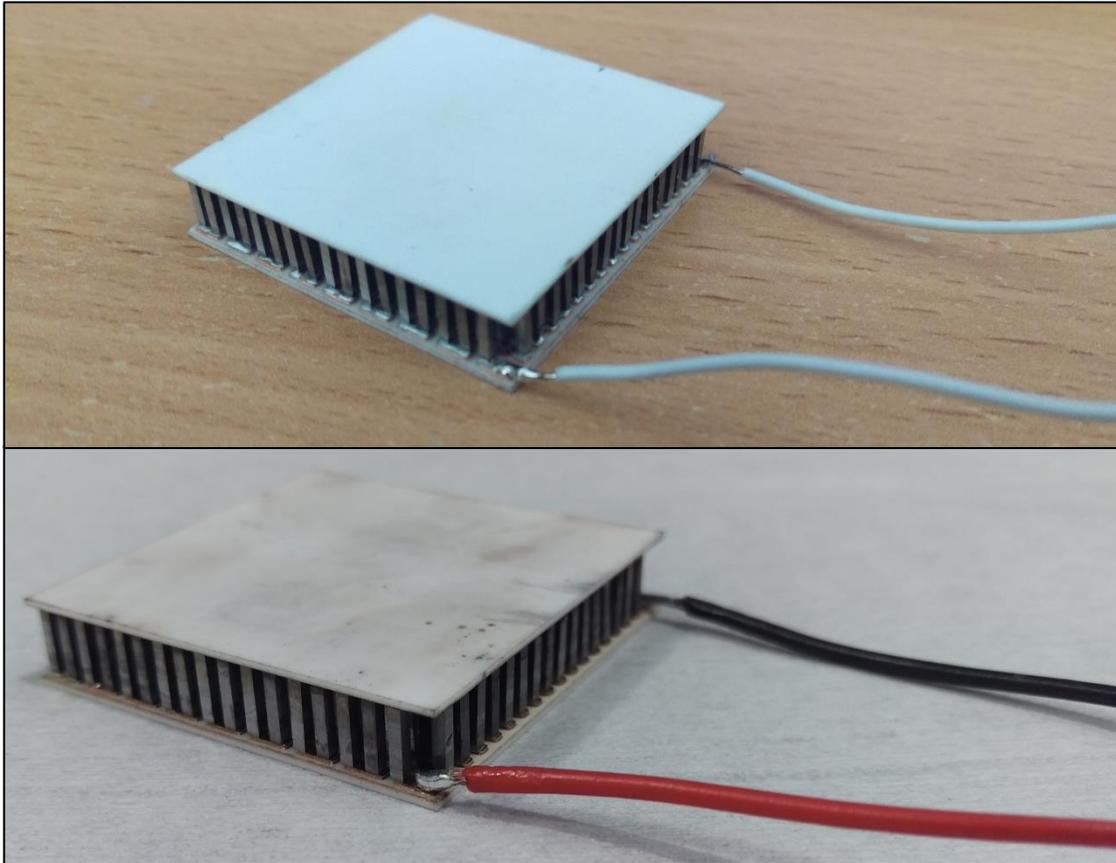


Add pellets



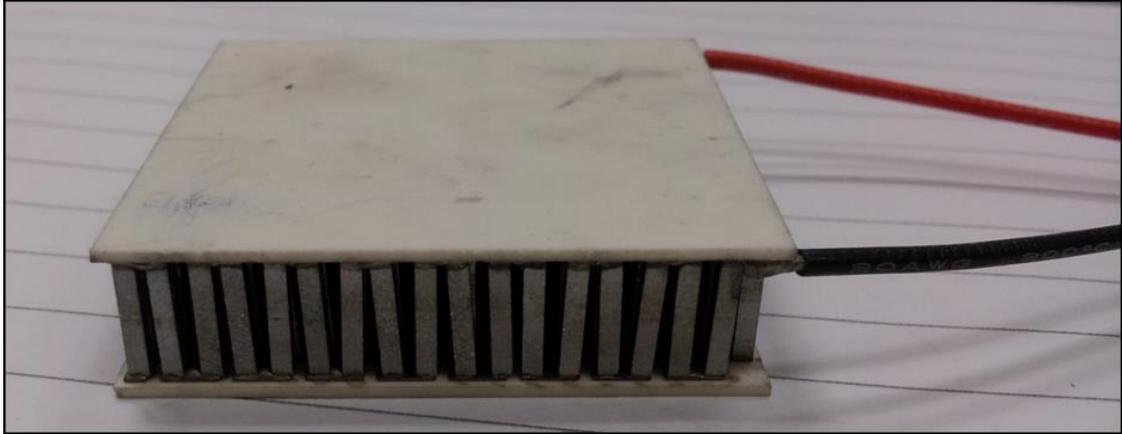
Add top ceramic
and wires

Modules for Space Exploration



- 161 couple devices.
- Pellet size: 1.2mm x 1.2mm x 6mm (l,w,h).
- 250°C hot side rating.
- 150°C cold side rating.
- UK manufactured.
- SPS material provided by Queen Mary University of London.

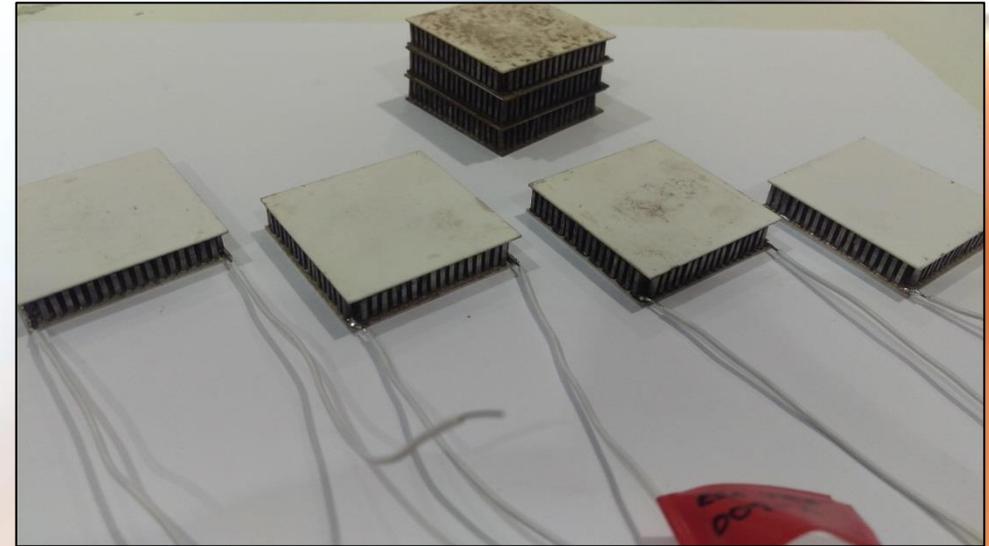
Custom Manufacturing for High AR



- Hard to maintain alignment with high aspect ratio pellets.
- Improvement of custom manufacturing methods.
- Commercial process to custom process shows improvement in pellet alignment.
- Improves:
 - Mechanically.
 - Electrically.
 - Thermally.

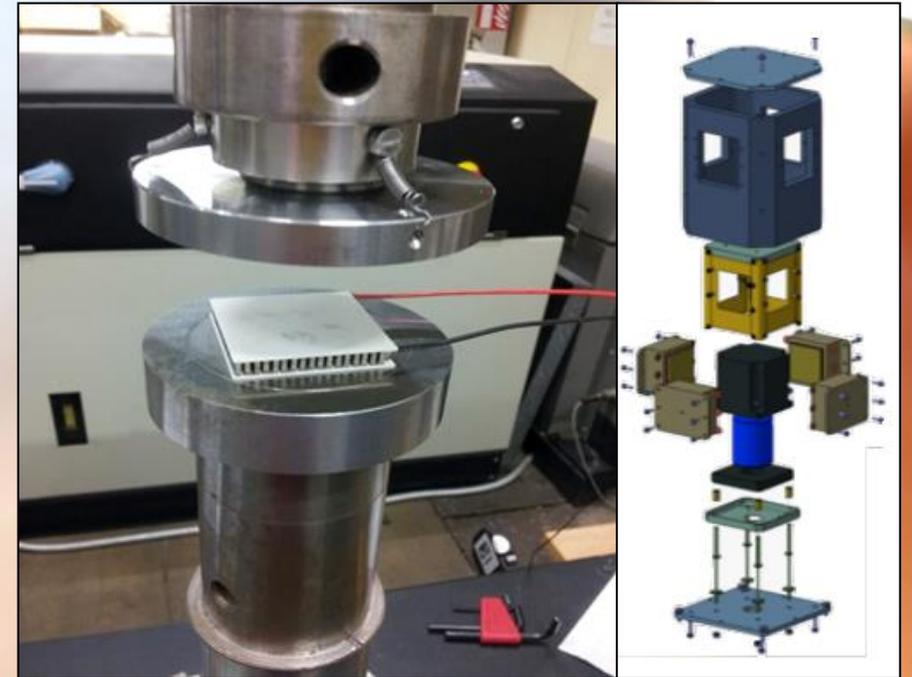
Module Manufacturing Quality

- Commercial n-type & SPS p-type Modules:
 - Average resistance = 17.44 Ω .
 - Standard Deviation = 2.50 Ω .
- SPS n-type & p-type:
 - Average resistance = 20.91 Ω .
 - Standard deviation = 0.89 Ω .
- Performance loss from n-type.
- Module manufacture is still challenging but build yield is at 75%.



Next Steps

- Modules are now with Leicester University.
- Mechanical testing and comparison to previous module builds.
- Thermoelectric generation performance testing on breadboard system.
- Explore routes for improving n-type ZT properties.

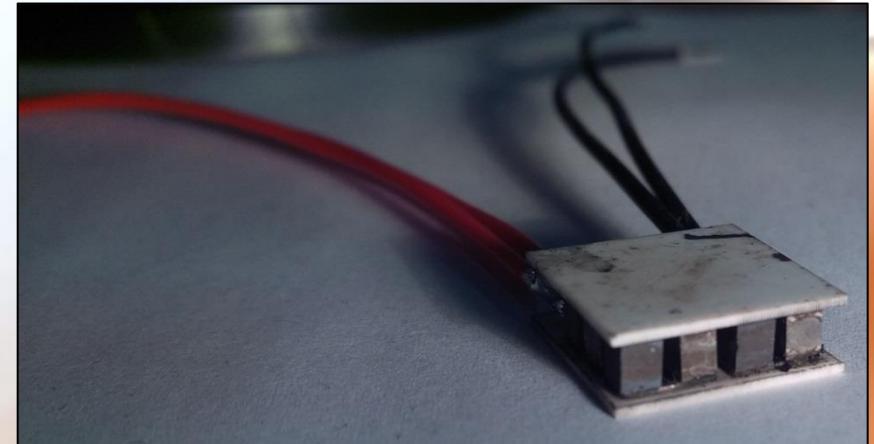


Research Update at ETL

- Currently concentrating on:
 - Skutterudite joining development and module manufacture.
 - Printed and flexible thermoelectrics.
 - PRINTEG pilot line for advanced module manufacture.
 - Silicide module development.

Research Update at ETL: Silicides

- Prototype silicide based modules manufactured.
 - 7-couple device.
 - Module XSec - 260mW/cm² @ dT =380K
 - Pellets XSec - 760mW/cm² @ dT =380K.
 - Estimated efficiency 2.8%.
- Current characterisation:
 - Performance data.
 - Reproducibility.
- Ongoing and into the future:
 - Lifetime testing and scalability.
 - Further testing in automotive system.
- More information to come at ECT 2017.



End & Thanks To Partners

